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Chinese Journal of Mechanical Engineering

# Effect of Degree-of-Symmetry on Kinetostatic Characteristics of Flexure Mechanisms: A Comparative Case Study

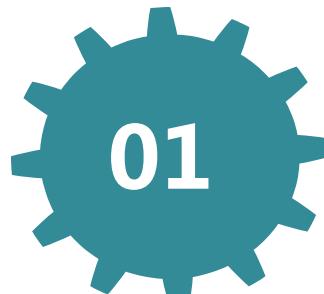
HE Xiaobing  
2017/11/20



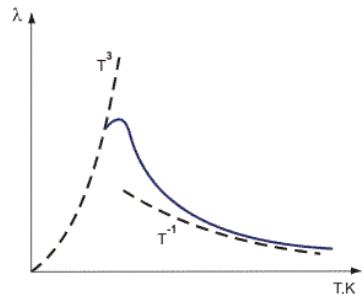
## Part1 : Question 😕



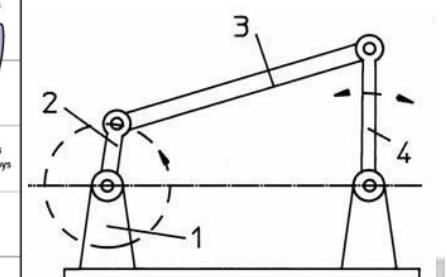
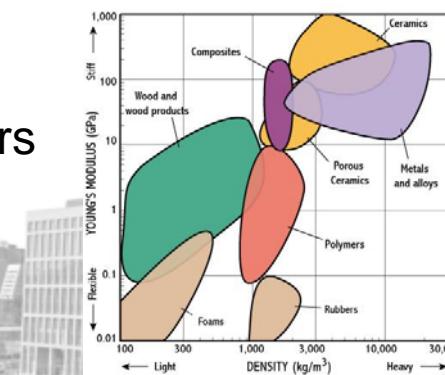
### ➤ Accuracy of flexure mechanisms is highly sensitive



External disturbances



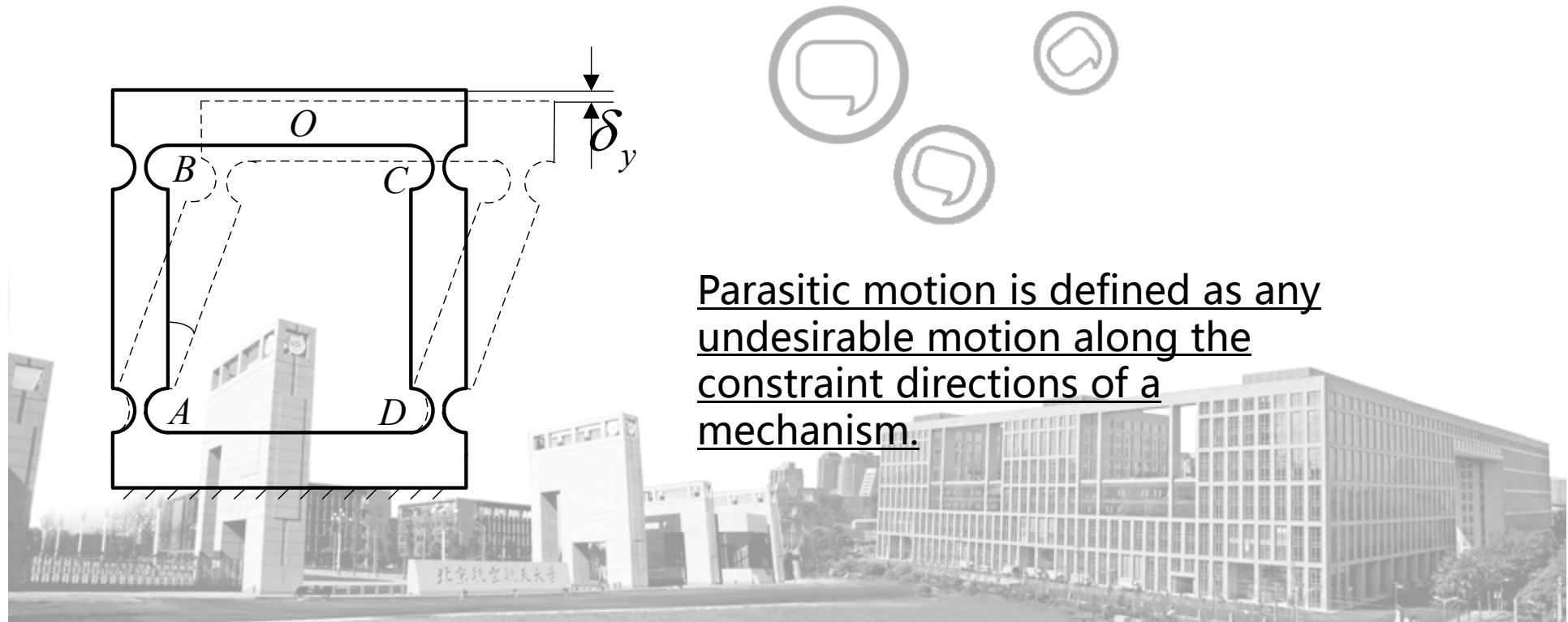
Internal factors



## Part1 : Question 😕



➤ Parasitic motion is an index for accuracy



Parasitic motion is defined as any undesirable motion along the constraint directions of a mechanism.

## Part1 : Question 😕



### ➤ Methods to reduce or eliminate the parasitic motion

parameter

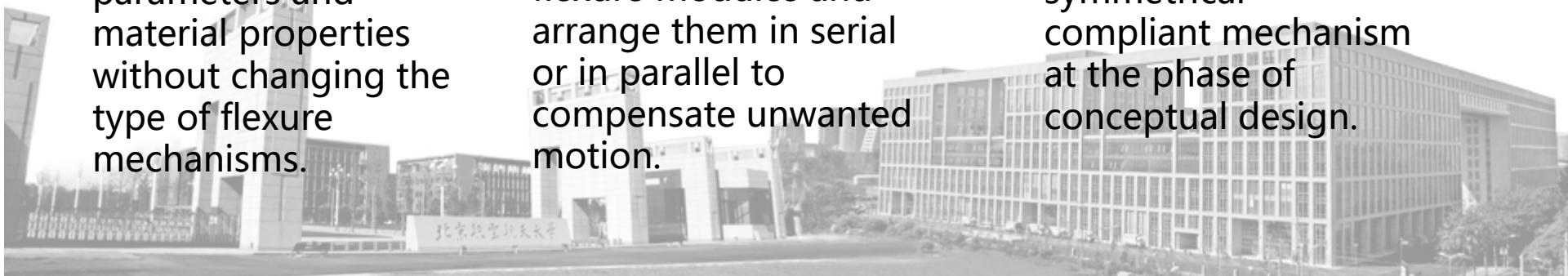
compensation

design

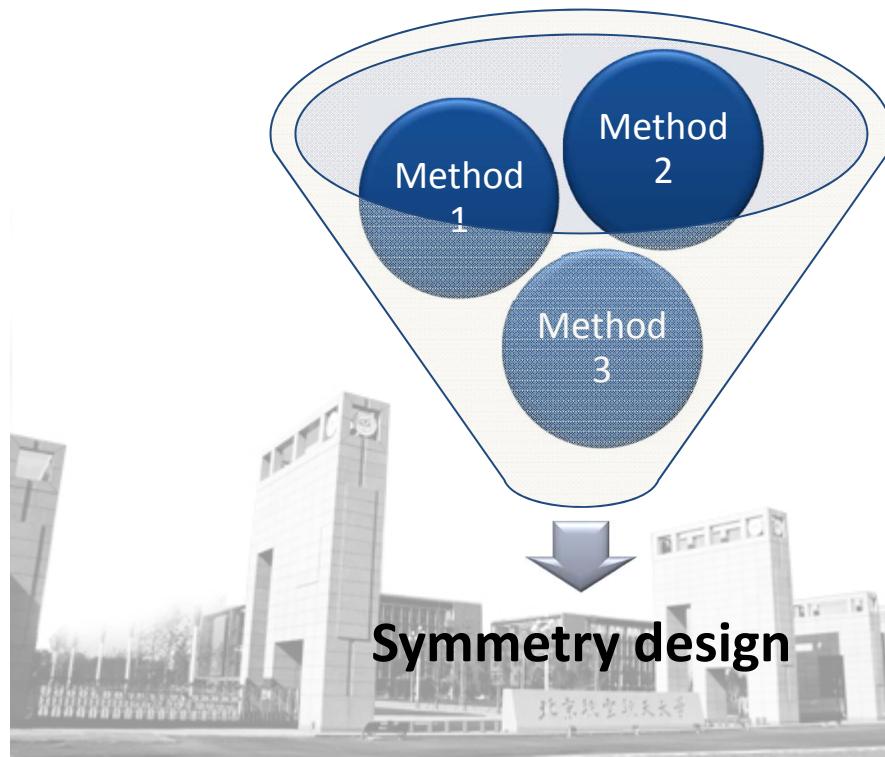
Tune the structural parameters and material properties without changing the type of flexure mechanisms.

Mirror two identical flexure modules and arrange them in serial or in parallel to compensate unwanted motion.

Design a practical full-symmetrical compliant mechanism at the phase of conceptual design.



# ➤ Common knowledge of symmetry



is it necessary to design the flexure mechanisms with symmetrical features as many as possible for better kinetostatic performance, when considering the resulting cost by the symmetry?

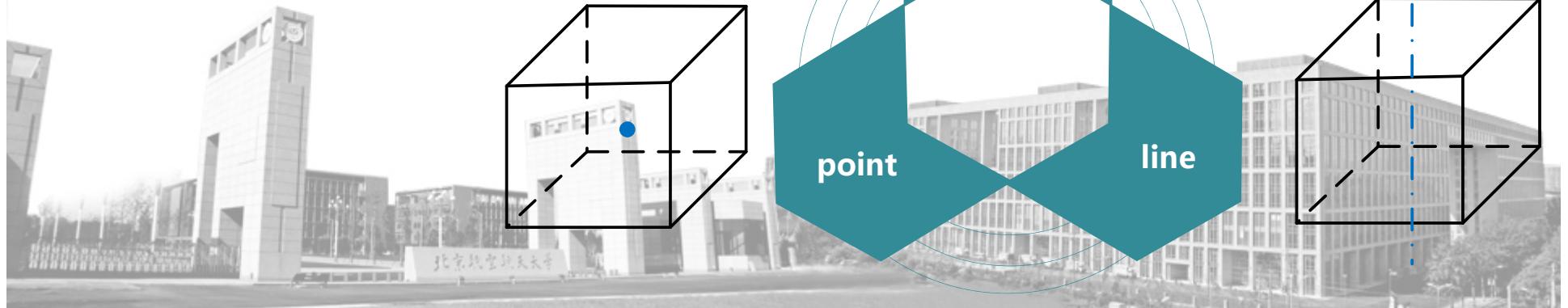


## Part2 : Analysis



### ➤ Degree of symmetry

Here, the Degree-of-Symmetry (DoS) is specifically constrained with plane symmetry.

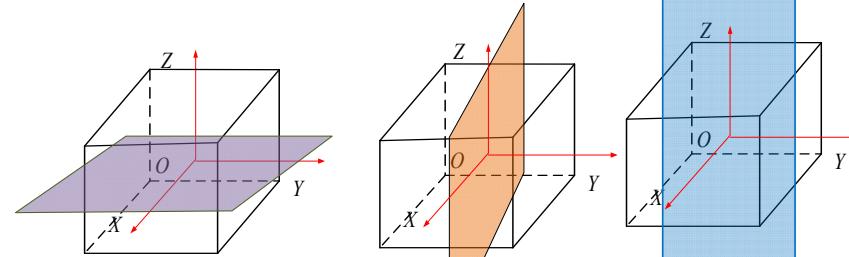


## Part2 : Analysis

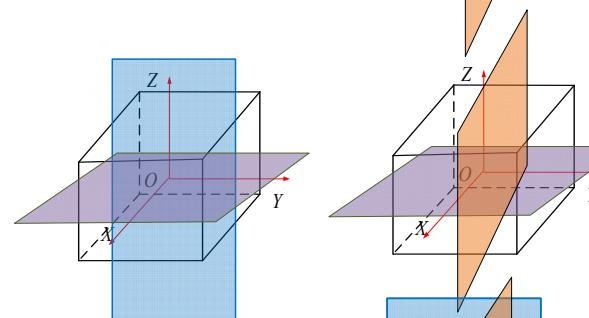


➤ **X-DoS**

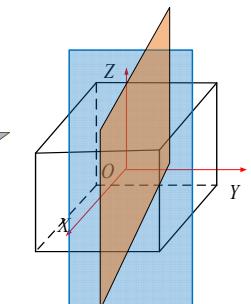
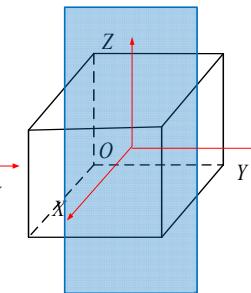
1-DoS



2-DoS



3-DoS

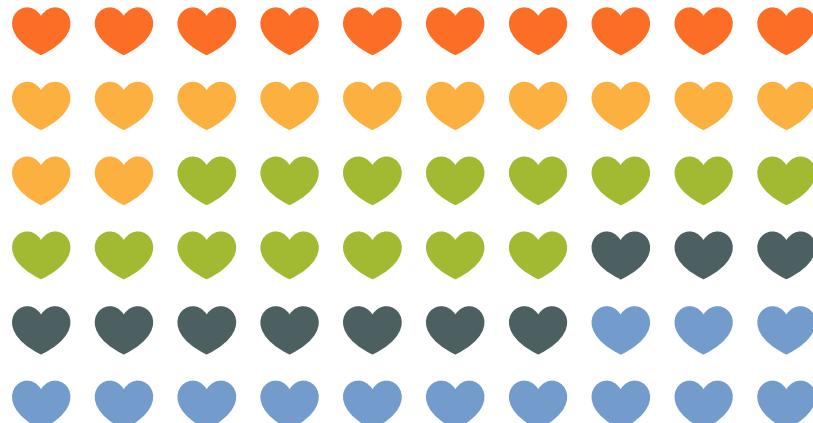


### ➤ A comparative study

So,  
what is the effect of DoS on  
kinetostatic characteristic of  
flexure mechanisms?



3-DoS > 2-DoS > 1-DoS ? ? ?



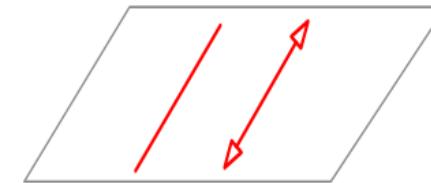
## Part2 : Analysis



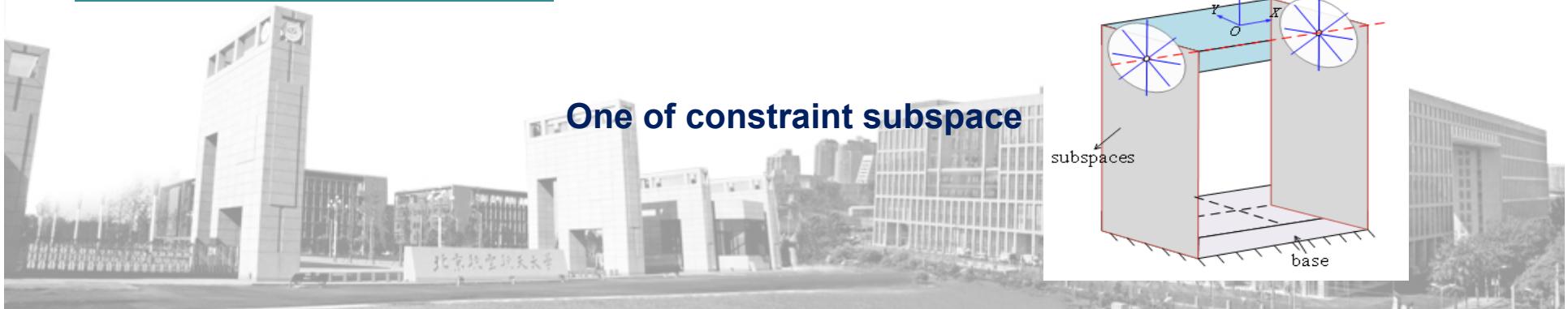
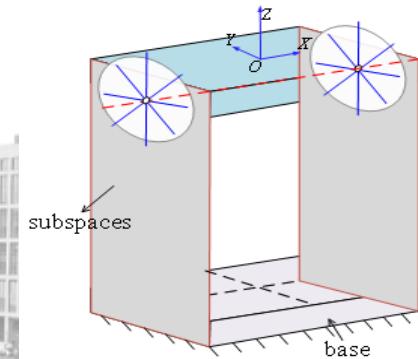
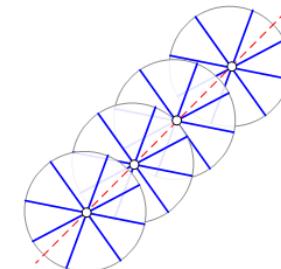
### ➤ Build the model

1R1T

The desired freedom space

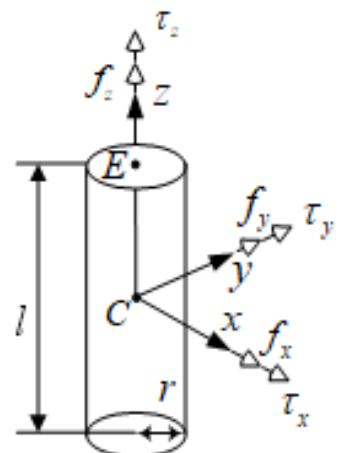


The desired constraint space



## Part2 : Analysis

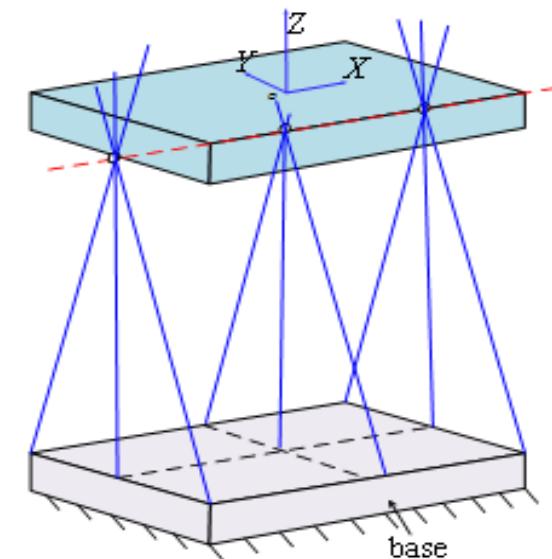
### ➤ Build the model



$$\frac{c_{cl1}}{c_{c33}} = \frac{c_{c22}}{c_{c33}} = \frac{l^3}{12EI_y} \sqrt{\frac{l}{EA}} = \frac{1}{3} \left(\frac{l}{r}\right)^2 = 133,$$

$$\frac{c_{c44}l^2}{c_{c33}} = \frac{c_{c55}l^2}{c_{c33}} = 1600, \quad \frac{c_{c66}l^2}{c_{c33}} = \frac{2l^2}{3(1+\mu)r^2} \geq 1000$$

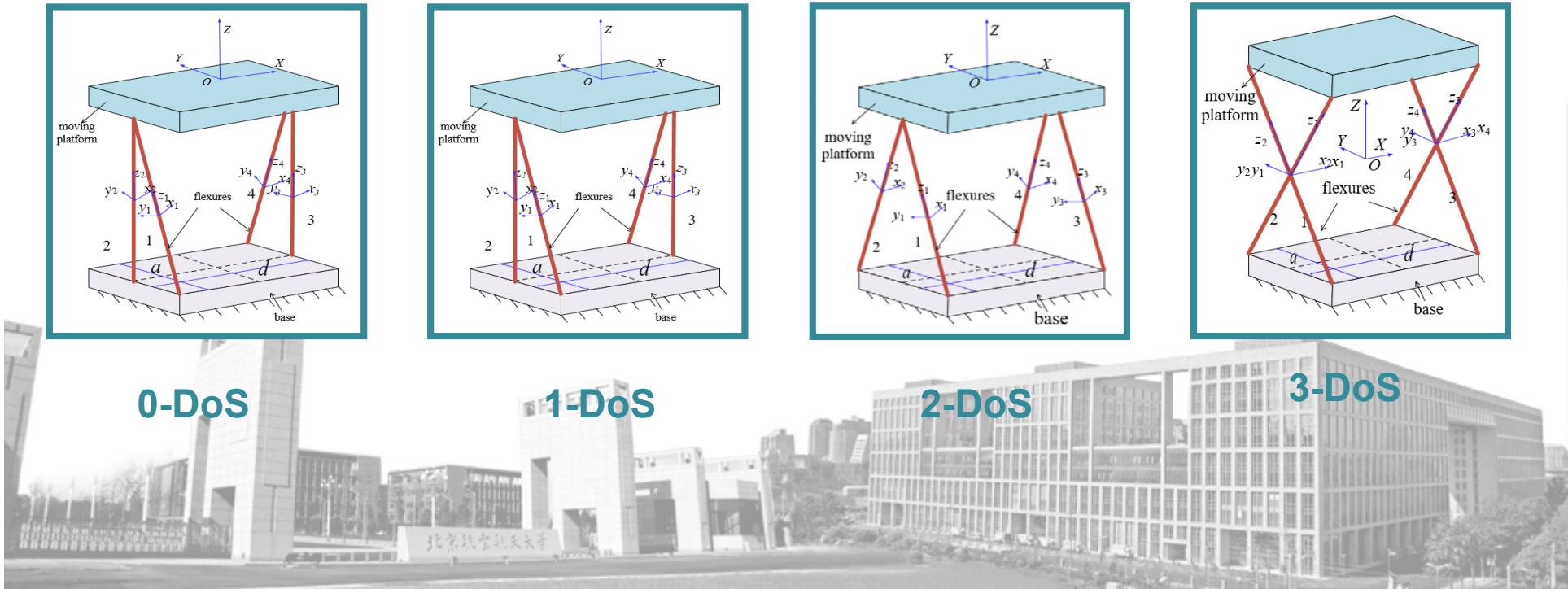
Each mechanism is a cylindrical joint that is composed of several identical beams distributed in two planes orthogonal to the direction of motion axes.



## Part2 : Analysis



### ➤ Build the model



# ➤ Overall compliance matrix

According to the screw theory and the theory of linear elasticity, the deformation denoted by the twist

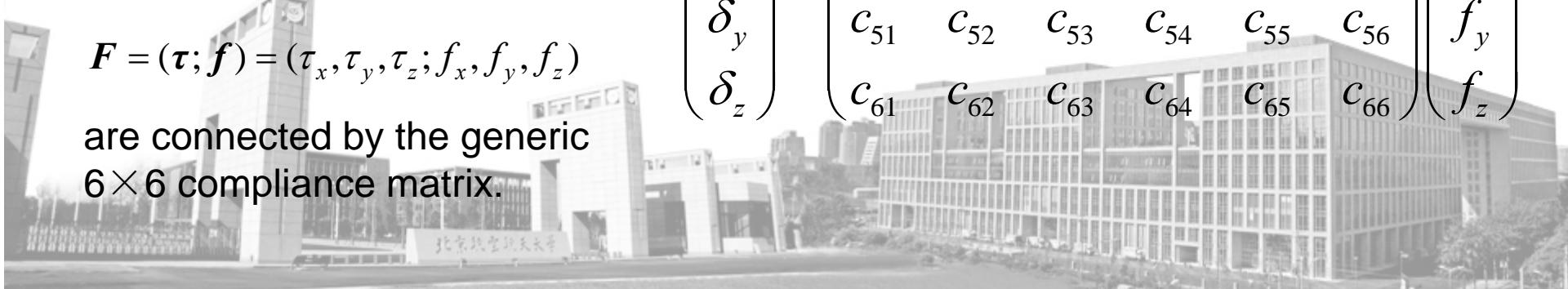
$$\xi = (\theta; \delta) = (\theta_x, \theta_y, \theta_z; \delta_x, \delta_y, \delta_z)$$

and the load wrench

$$F = (\tau; f) = (\tau_x, \tau_y, \tau_z; f_x, f_y, f_z)$$

are connected by the generic  $6 \times 6$  compliance matrix.

$$\begin{pmatrix} \theta_x \\ \theta_y \\ \theta_z \\ \delta_x \\ \delta_y \\ \delta_z \end{pmatrix} = \begin{pmatrix} c_{11} & c_{12} & c_{13} & c_{14} & c_{15} & c_{16} \\ c_{21} & c_{22} & c_{23} & c_{24} & c_{25} & c_{26} \\ c_{31} & c_{32} & c_{33} & c_{34} & c_{35} & c_{36} \\ c_{41} & c_{42} & c_{43} & c_{44} & c_{45} & c_{46} \\ c_{51} & c_{52} & c_{53} & c_{54} & c_{55} & c_{56} \\ c_{61} & c_{62} & c_{63} & c_{64} & c_{65} & c_{66} \end{pmatrix} \begin{pmatrix} \tau_x \\ \tau_y \\ \tau_z \\ f_x \\ f_y \\ f_z \end{pmatrix}$$



### ➤ Overall compliance matrix

Only a force  $f_x$   
is imposed on  
the mobile  
platform.

$$\begin{pmatrix} \theta_x \\ \theta_y \\ \theta_z \\ \delta_x \\ \delta_y \\ \delta_z \end{pmatrix} = \begin{pmatrix} c_{11} & c_{12} & c_{13} & c_{14} & c_{15} & c_{16} \\ c_{21} & c_{22} & c_{23} & c_{24} & c_{25} & c_{26} \\ c_{31} & c_{32} & c_{33} & c_{34} & c_{35} & c_{36} \\ c_{41} & c_{42} & c_{43} & c_{44} & c_{45} & c_{46} \\ c_{51} & c_{52} & c_{53} & c_{54} & c_{55} & c_{56} \\ c_{61} & c_{62} & c_{63} & c_{64} & c_{65} & c_{66} \end{pmatrix} \begin{pmatrix} \tau_x \\ \tau_y \\ \tau_z \\ f_x \\ f_y \\ f_z \end{pmatrix}$$



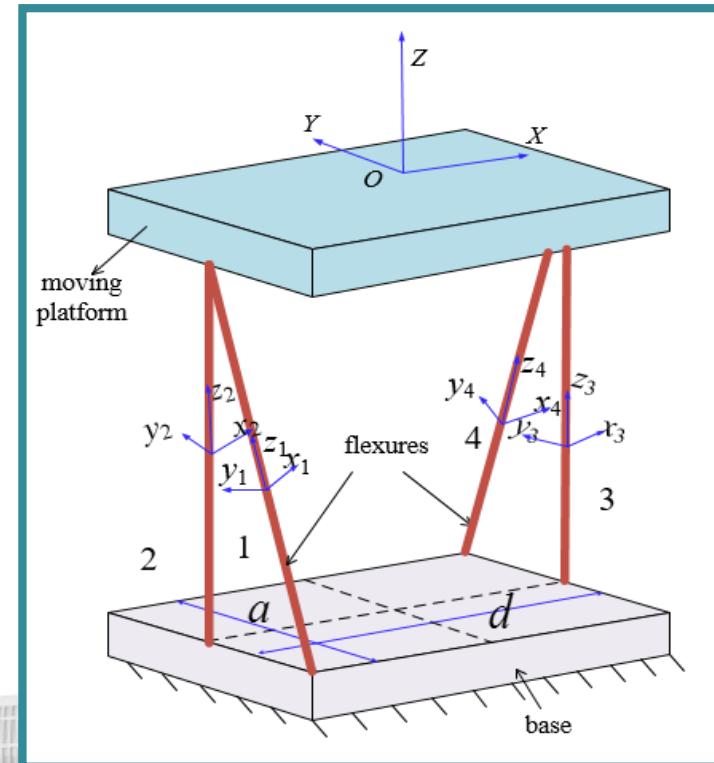
## Part2 : Analysis

### ➤ Parasitic motion error analysis

$$C_{0-\text{DOS}} = \begin{pmatrix} c_{11} & c_{12} & 0 & c_{14} & c_{15} & 0 \\ c_{21} & c_{22} & 0 & c_{24} & c_{25} & 0 \\ 0 & 0 & c_{33} & 0 & 0 & c_{36} \\ c_{41} & c_{42} & 0 & c_{44} & c_{45} & 0 \\ c_{51} & 0 & c_{53} & c_{54} & c_{55} & 0 \\ 0 & 0 & c_{63} & 0 & 0 & c_{66} \end{pmatrix}$$



$$\xi_0 = \theta_x + \theta_y + \delta_x + \delta_y = c_{14}f_x + c_{24}f_x + c_{44}f_x + c_{54}f_x$$



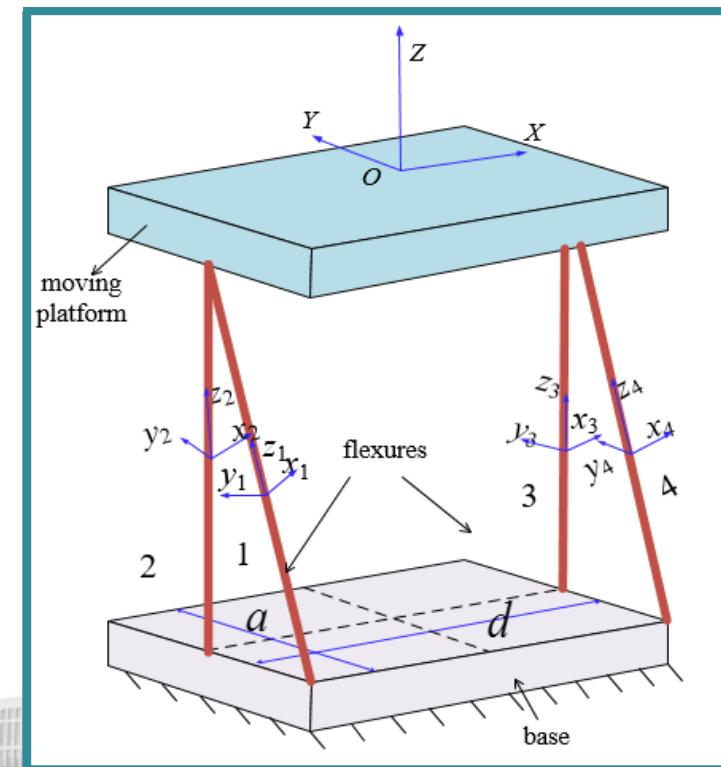
## Part2 : Analysis

### ➤ Parasitic motion error analysis

$$C_{1-\text{DOS}} = \begin{pmatrix} c_{11} & 0 & 0 & 0 & c_{15} & c_{16} \\ 0 & c_{22} & c_{23} & c_{24} & 0 & 0 \\ 0 & c_{32} & c_{33} & c_{34} & 0 & 0 \\ 0 & c_{42} & c_{43} & c_{44} & 0 & 0 \\ c_{51} & 0 & 0 & 0 & c_{55} & c_{56} \\ c_{61} & 0 & 0 & 0 & c_{65} & c_{66} \end{pmatrix}$$



$$\xi_1 = \theta_y + \theta_z + \delta_x = c_{24}f_x + c_{34}f_x + c_{44}f_x$$



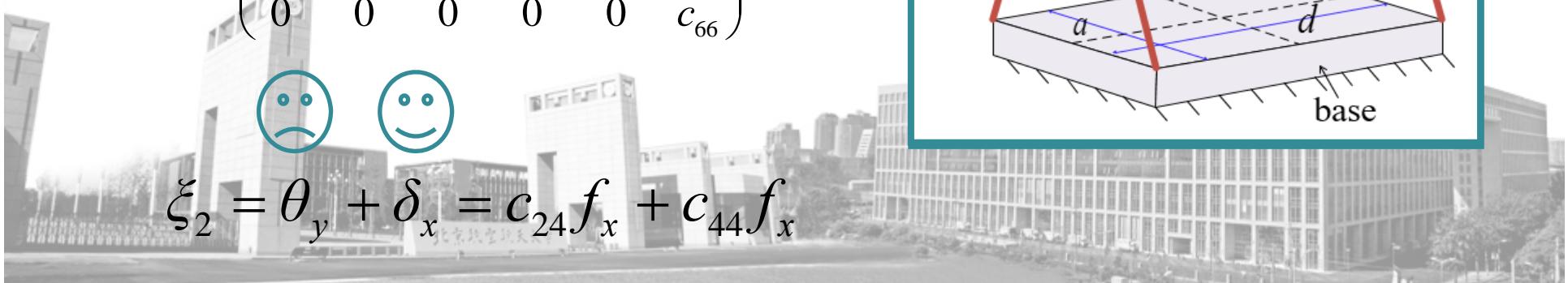
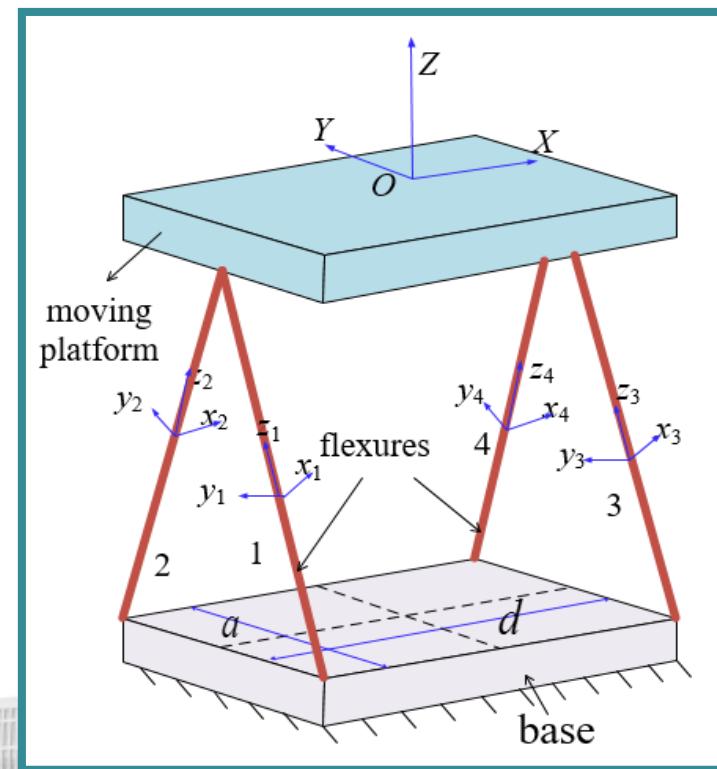
## Part2 : Analysis

### ➤ Parasitic motion error analysis

$$C_{2-\text{DOS}} = \begin{pmatrix} c_{11} & 0 & 0 & 0 & c_{15} & 0 \\ 0 & c_{22} & 0 & c_{24} & 0 & 0 \\ 0 & 0 & c_{33} & 0 & 0 & 0 \\ 0 & c_{42} & 0 & c_{44} & 0 & 0 \\ c_{51} & 0 & 0 & 0 & c_{55} & 0 \\ 0 & 0 & 0 & 0 & 0 & c_{66} \end{pmatrix}$$



$$\xi_2 = \theta_y + \delta_x = c_{24}f_x + c_{44}f_x$$

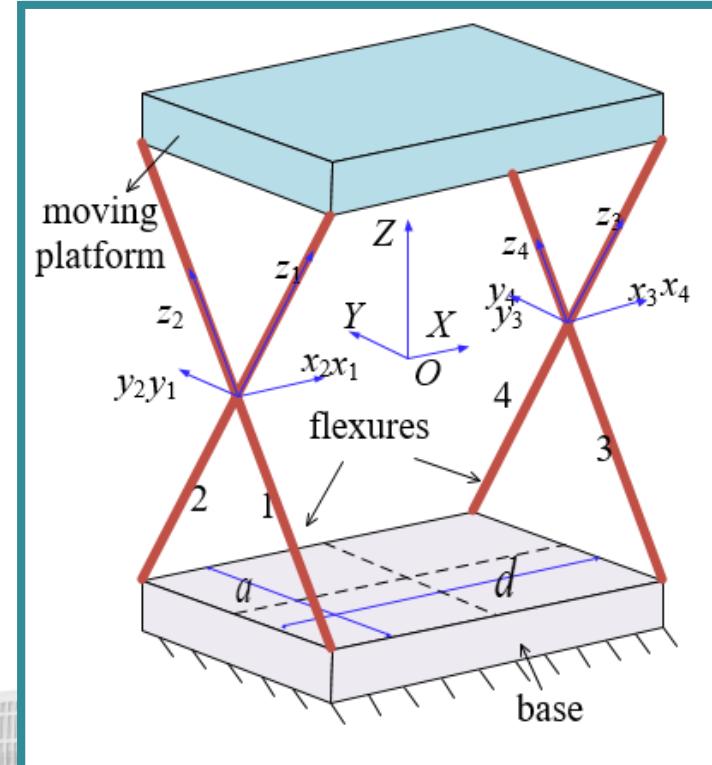


## Part2 : Analysis

### ➤ Parasitic motion error analysis

$$C_{3-\text{DOS}} = \begin{pmatrix} c_{11} & 0 & 0 & 0 & 0 & 0 \\ 0 & c_{22} & 0 & 0 & 0 & 0 \\ 0 & 0 & c_{33} & 0 & 0 & 0 \\ 0 & 0 & 0 & c_{44} & 0 & 0 \\ 0 & 0 & 0 & 0 & c_{55} & 0 \\ 0 & 0 & 0 & 0 & 0 & c_{66} \end{pmatrix}$$

$$\xi_3 = \delta_x = c_{44} f_x$$



## Part2 : Analysis



➤ **The best design**

3-DoS type  
Because of all symmetric planes



## Part3 : Verification



### ➤ Finite element analysis

ANSYS Rigid platform: SOLID 186  
15.0

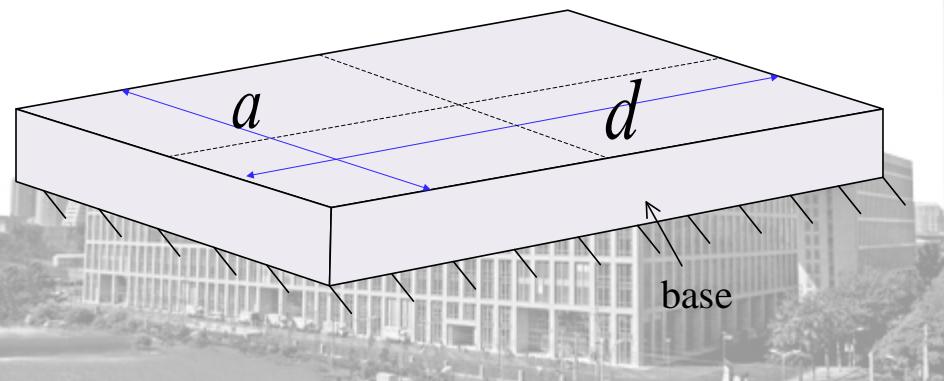
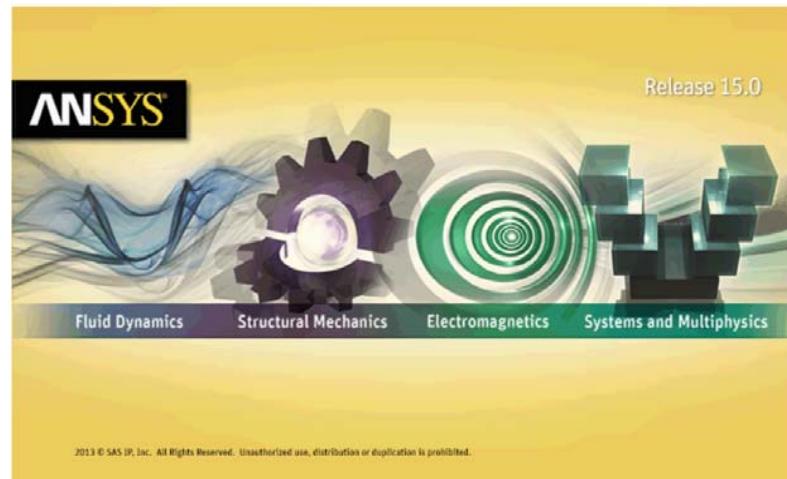
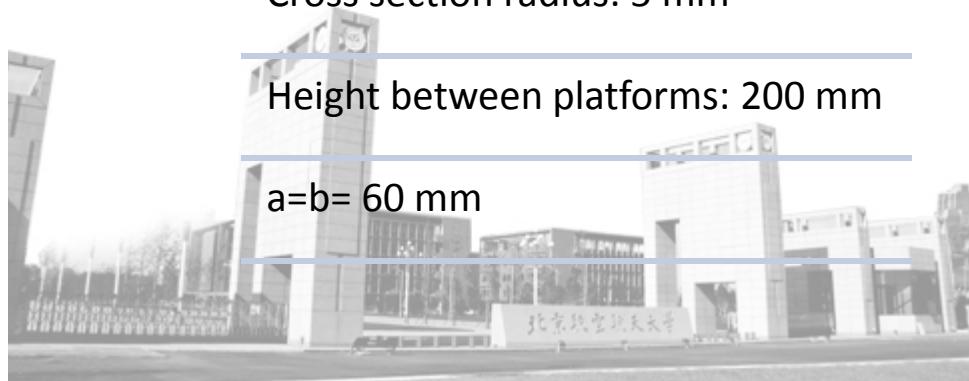
Flexure beam: BEAM 189

$E=70 \text{ GPa}$ ,  $\mu=0.34$

Cross section radius: 5 mm

Height between platforms: 200 mm

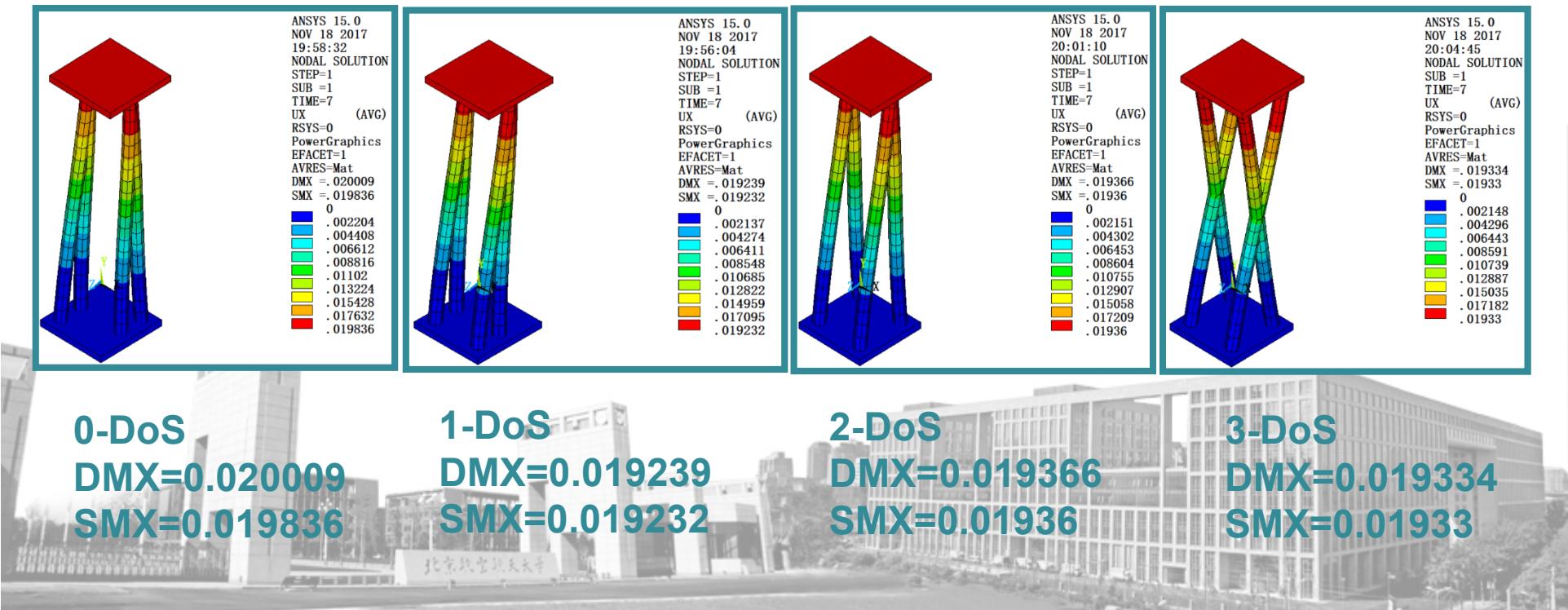
$a=b=60 \text{ mm}$



## Part3 : Verification



### ➤ FEA simulations of each mechanism



### ➤ Optimize the 3-DoS flexure mechanism

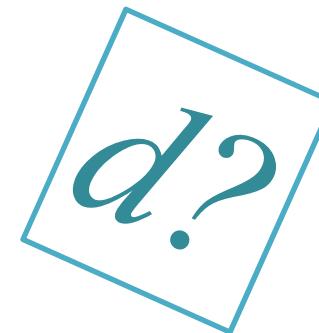
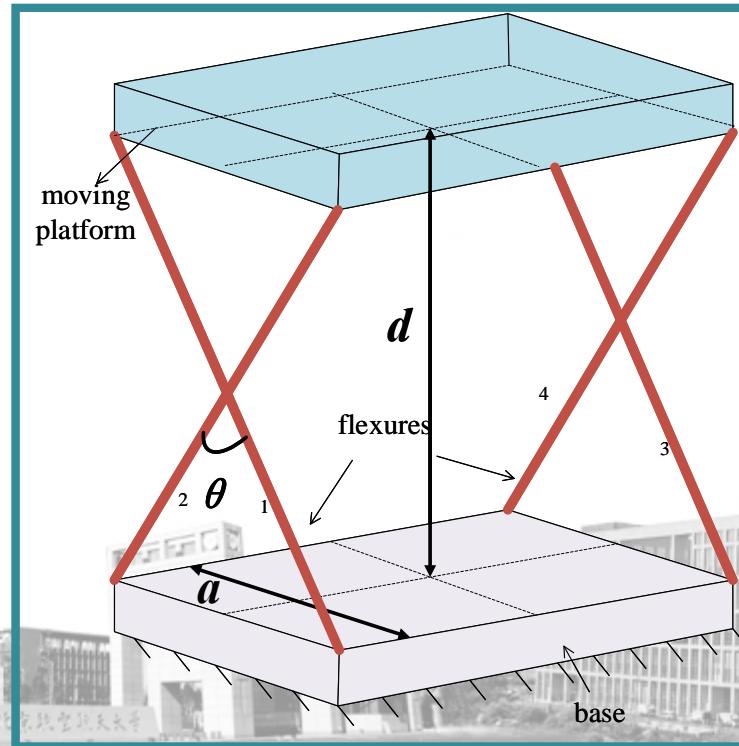
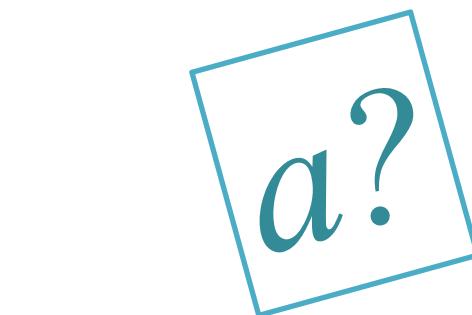


The entry  $c_{11}$  and  $c_{44}$  should be the dominant ones for ensuring a rotation along the x-axis and a translational about x-axis.

$$\left\{ \begin{array}{l} c_{11} = \frac{a}{E\pi r^4 \sin \theta} \\ c_{22} = \frac{a^3}{\pi r^2 \sin \theta (2Ga^2r^2 \sin^2 \theta + 12Ed^2r^2 \sin^4 \theta + Ea^2r^2 \cos^2 \theta + 4Ea^2d^2 \cos^2 \theta)} \\ c_{33} = \frac{a^3}{\pi r^2 \sin \theta (2Ga^2r^2 \cos^2 \theta + Ea^2r^2 \sin^2 \theta + 12Ed^2r^2 \cos^2 \theta \sin^2 \theta + 4Ea^2d^2 \sin^2 \theta)} \\ c_{44} = \frac{a^3}{12E\pi r^4 \sin^3 \theta} \\ c_{55} = \frac{a^3}{4E\pi r^2 \sin^3 \theta (3r^2 \cos^2 \theta + a^2)} \\ c_{66} = \frac{a^3}{4E\pi r^2 \sin \theta (3r^2 \sin^4 \theta + a^2 \cos^2 \theta)} \end{array} \right.$$

## Part4 : Optimization

### ➤ Optimize the 3-DoS flexure mechanism



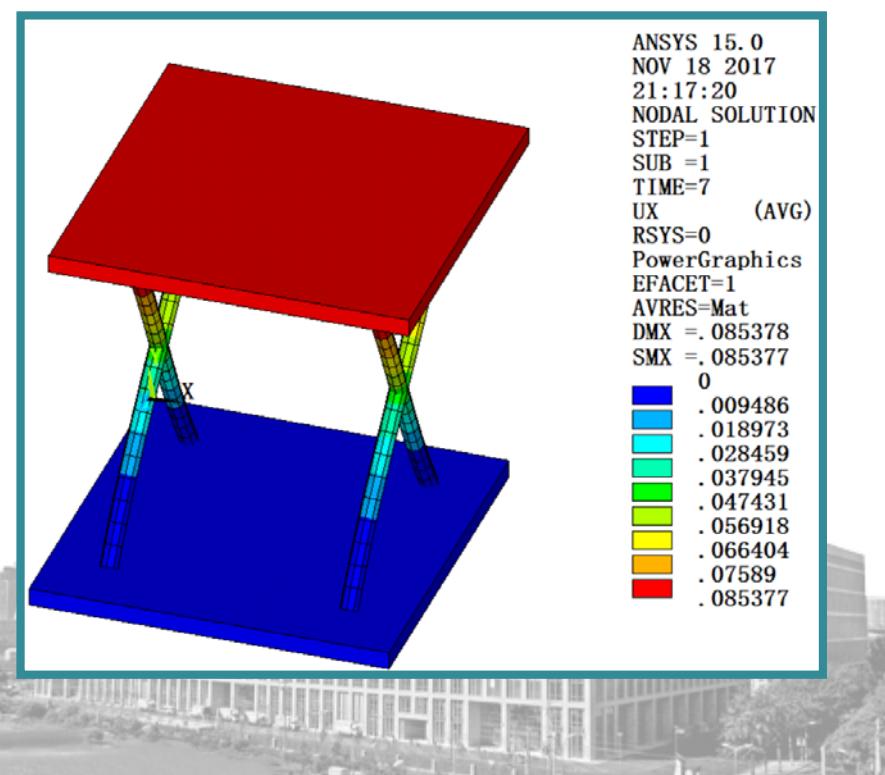
## Part4 : Optimization



### ➤ Optimize the 3-DoS flexure mechanism

Optimal parameters for 3-DoS flexure model

Beam orientation ( $\theta$ )	$\theta=\pi/4$
Two end points distance (a)	200 mm
Interval distance (d)	200 mm
Radius of beam (r)	5 mm



## Part5 : Discussion

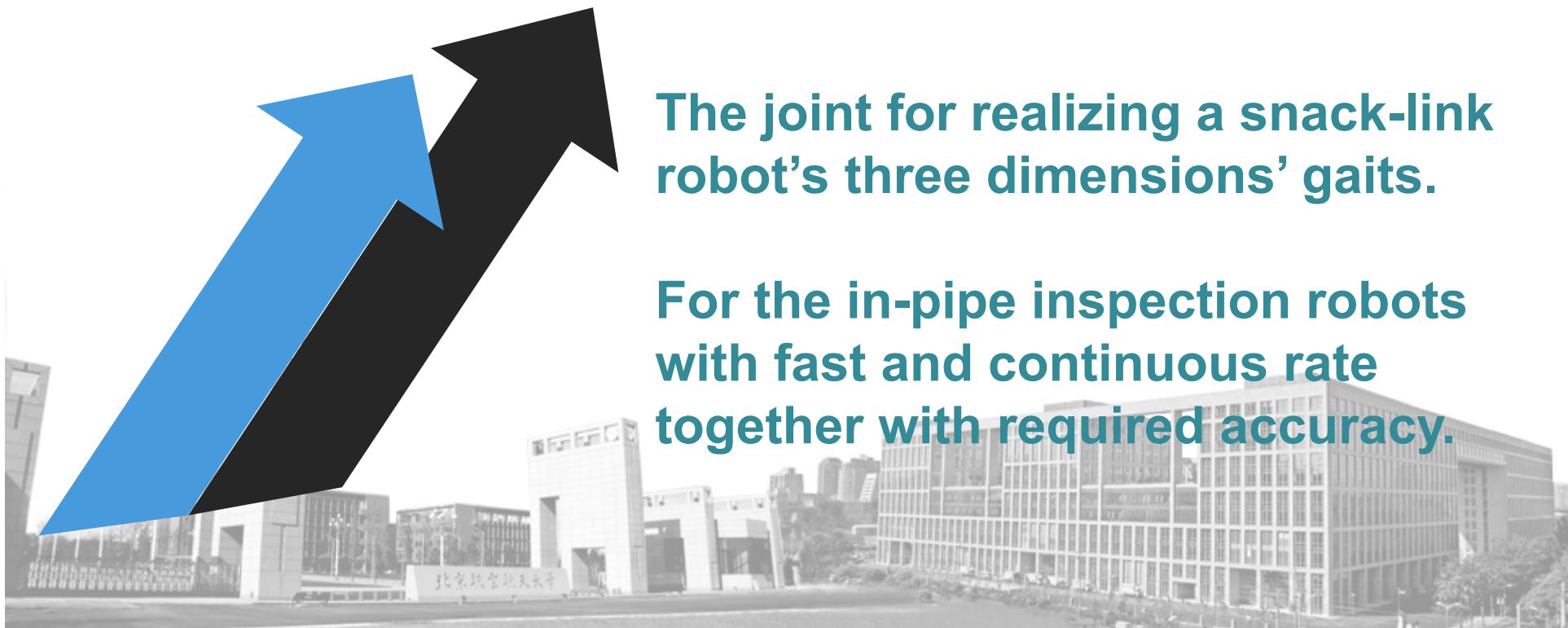


➤ Larger DoS, better result.

In the design process, it is indeed better to generate as many DoS as possible from the very beginning. With the different DoS, people can further adopt certain methods to alleviate the unwanted parasitic motion.



### ➤ Application of flexure mechanisms with cylindrical motion



## Part5 : Conclusion



01

Design

Using FACT method to design flexure mechanisms with different number of symmetric planes.

02

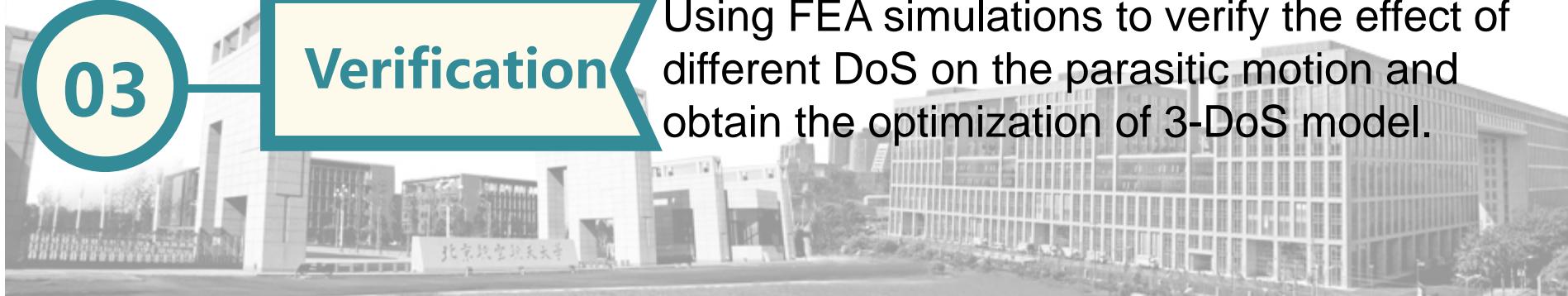
Analysis

Using overall compliance matrix to analyze the effect of symmetrical geometry on the kinetostatic characteristics.

03

Verification

Using FEA simulations to verify the effect of different DoS on the parasitic motion and obtain the optimization of 3-DoS model.



## Acknowledge



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THANK YOU